

Roll No.

Total No. of Pages : 02

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B.Tech.(ANE)/(Aerospace Engg.) (2012 Onwards) (Sem.-3)

**AERODYNAMICS- I**

Subject Code : ANE-203

M.Code : 60502

Time : 3 Hrs.

Max. Marks : 60

**INSTRUCTIONS TO CANDIDATES :**

1. SECTION-A is COMPULSORY consisting of TEN questions carrying TWO marks each.
2. SECTION-B contains FIVE questions carrying FIVE marks each and students have to attempt any FOUR questions.
3. SECTION-C contains THREE questions carrying TEN marks each and students have to attempt any TWO questions.

**SECTION-A**

1. Answer briefly :

- (a) Pathline
- (b) Potential flow
- (c) Conservation equations
- (d) Centre of pressure
- (e) Aerodynamic force
- (f) Lift coefficient
- (g) Pressure coefficient
- (h) Steady-incompressible flow
- (i) Two dimensional flow separation
- (j) Boundary Layer

**SECTION-B**

2. Define circulation, angular velocity, vorticity and time rate of strain of a fluid element in three-dimensional space.
3. Starting from the first principle. Derive Euler Equation of motion.
4. Describe Doublet flow. Derive the velocity potential and stream function for it.
5. Explain, why a laminar boundary layer separates more easily than a turbulent boundary layer.
6. Derive an expression for velocity potential and stream function for a vortex flow.

**SECTION-C**

7. Show that the pressure coefficient around a circular cylinder placed in an ideal, incompressible flow is given by :  $C_p = 1 - 4 \sin^2 \theta$ . Where 'θ' is the angular coordinate on the surface of the cylinder measured from the lagging stagnation point.
8. Derive an expression for 2-D continuity equation applicable to compressible, viscous, non-uniform and unsteady flows. What are the advantages of integral form of continuity equation over the differential form?
9. Write short notes on the followings :
  - (a) Kantian's Integral equation
  - (b) Blasius solution for a flat plate

**NOTE : Disclosure of Identity by writing Mobile No. or Making of passing request on any page of Answer Sheet will lead to UMC against the Student.**