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B.Tech.(ME) (2011 Onwards) (Sem.-4)
APPLIED THERMODYNAMICS-II

Subject Code : BTME-404 M.Code : 59132

Time: 3 Hrs. Max. Marks: 60

INSTRUCTIONS TO CANDIDATES:

- SECTION-A is COMPULSORY consisting of TEN questions carrying TWO marks each.
- SECTION-B contains FIVE questions carrying FIVE marks each and students have to attempt any FOUR questions.
- SECTION-C contains THREE questions carrying TEN marks each and students have to attempt any TWO questions.

SECTION-A

Answer briefly :

- a) Write the different applications of compressed air.
- b) List the advantages of multistage compression.
- c) Define 'degree of reaction' in relation to axial flow compressor.
- d) What is surging?
- e) Write the merits and demerits of axial flow compressor.
- Mention the application of gas turbine.
- g) Name the various modes of energy transfer in impeller.
- List the requirements of a gas turbine combustion chamber.
- i) Define 'angle of attack'.
- Write name of different fuels used in the gas turbine.

 $10 \times 2 = 20$

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SECTION-B

- Explain with a neat sketch the working of a single stage reciprocating compressor.
- Distinguish between positive and non-positive displacement type compressors.
- Comment on the validity of the statement: 'The thermodynamic efficiency of a centrifugal compressor is referred to adiabatic condition where as that of reciprocating compressor is referred to isothermal condition'.
- A jet propulsion system has to create a thrust of 90 tonnes to move the system at a velocity of 650 km/hr. Find the exit gas velocity and propulsion efficiency if the gas flow rate through the system is restricted to 25 kg/s.
- State the difference between the jet propulsion and rocket propulsion system.

SECTION-C

Set up the following expression for volumetric efficiency with respect to free air delivery,

$$\eta_{vol} = \frac{p_1 T_0}{p_0 T_1} \left[1 + C - C \left(\frac{p_2}{p_1} \right)^{1/\sigma} \right]$$

Where the symbols in the expression have their usual meanings.

8. A two-cylinder double acting reciprocating compressor sucks in air at pressure 0.98 bar and temperature 300 K. The delivery of the compressed air to the receiver is at 6 bar. The breathing capacity of the compressor is stated to be 2.5 m³/min when measured at 1 bar and 228 K. The mean speed of the compressor is limited to 120 mm/minute and the stroke is 0.75 times the cylinder diameter. If the law of compression is pV¹.³, make calculation for (a) Cylinder diameter and stroke length (b) Compressor speed in rev/min (c) Shaft power if the mechanical efficiency is 85% (d) isothermal efficiency.

Neglect the effect of clearance and piston rod diameter.

 Sketch Brayton cycle on p-v and T-s plots and derive a relation for its thermal efficiency in terms of pressure ratio.

NOTE: Disclosure of Identity by writing Mobile No. or Making of passing request on any page of Answer Sheet will lead to UMC against the Student.

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