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Total No. of Pages : 03

Total No. of Questions : 09

**B.Tech.(ANE) (Sem.-4)**  
**AIRCRAFT STRUCTURES – I**

Subject Code : ANE-206

M.Code : 60514

Time : 3 Hrs.

Max. Marks : 60

**INSTRUCTIONS TO CANDIDATES :**

1. SECTION-A is COMPULSORY consisting of TEN questions carrying TWO marks each.
2. SECTION-B contains FIVE questions carrying FIVE marks each and students have to attempt any FOUR questions.
3. SECTION-C contains THREE questions carrying TEN marks each and students have to attempt any TWO questions.

**SECTION-A****1. Write briefly :**

- a) Derive the relations between stress and strain in a plane stress problem.
- b) State the principle of superposition.
- c) What is Prandtl stress function solution?
- d) Derive an expression for the strain energy of a bar subjected to uniaxial tensile force.
- e) What is Shear lag?
- f) What is meant by a doubly redundant structure?
- g) State the different types of loads acting on aircraft structural components.
- h) Write down an expression for the shear flow in an open section beam subjected to shear forces  $S_x$  and  $S_y$ .
- i) Explain the term : 'Margin of Safety'
- j) What are principal planes and principal stresses?



### SECTION-B

2. A point in a strained material is subjected to a tensile stress of  $120 \text{ N/mm}^2$  and a compressive stress of  $80 \text{ N/mm}^2$  acting at right angles to each other. Find the normal stress, tangential stress and the resultant stress on a plane inclined at an angle of  $30^\circ$  with the compressive stress.
3. A circular section bar of radius  $R$  is subjected to equal and opposite torque  $T$  at each of its free ends. Determine ; (a) rate of twist and (b) stress distribution.
4. Calculate the rotation of point 1 of the cantilever beam in Fig. 1. Using the principle of complementary virtual work. Neglect shear deflection.

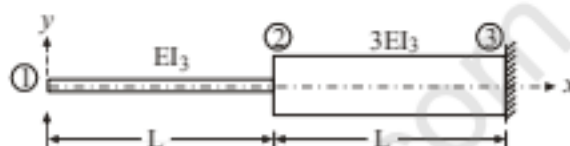


Fig.1

5. Sketch a typical V-H diagram and briefly explain (a) stall regions (b) method of plotting various segments of the diagram (c) Red-line speed of the air plane and (d) Significance of maneuver point and corner velocity.
6. Describe the different types of fitting failures with the help of suitable sketches.

### SECTION-C

7. A cantilever of length  $L$  and depth  $2h$  is in a state of plane stress. The cantilever is of unit thickness, is rigidly supported at the end  $x = L$  and is loaded as shown in fig. 2.

Show that the stress function :

$$\phi = A x^3 + B x^2 y + C y^3 = D (5x^2 y^3 - y^5)$$

is valid for the beam and evaluate the constants  $A$ ,  $B$ ,  $C$  and  $D$  in terms of  $q$  and  $h$ .

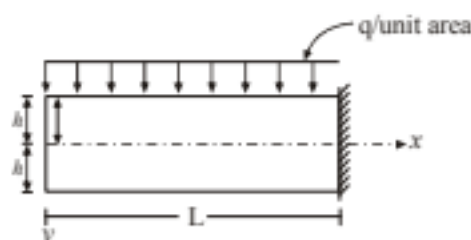


Fig.2

8. A structure is shown in Fig. 3. Assuming the member to be of uniform cross-section throughout find :
- Strain energy stored by the structure and hence
  - Determine the vertical deflection of end A.

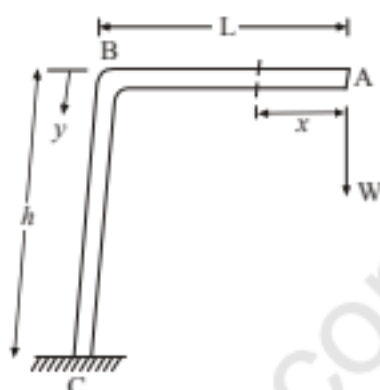


Fig.3

9. Write notes on :
- Maneuver Loads
  - Direct stress distribution due to bending.

**NOTE : Disclosure of Identity by writing Mobile No. or Making of passing request on any page of Answer Sheet will lead to UMC against the Student.**