

**Total No. of Pages : 03**

**Total No. of Questions : 09**

**B.Tech.(Aerospace Engg.) (2012 Batch) (Sem.-6)**

# FINITE ELEMENT METHODS

**Subject Code : ASPE-313**

**M.Code : 72458**

**Time : 3 Hrs.**

**Max. Marks : 60**

### INSTRUCTIONS TO CANDIDATES :

1. **SECTION-A is COMPULSORY consisting of TEN questions carrying TWO marks each.**
2. **SECTION-B contains FIVE questions carrying FIVE marks each and students have to attempt any FOUR questions.**
3. **SECTION-C contains THREE questions carrying TEN marks each and students have to attempt any TWO questions.**
4. **Assume suitably missing data if any.**

## SECTION-A

- 1. Answer briefly :**
- Write the expression of six strain components in terms of displacements.
  - Differentiate between natural and Cartesian coordinate systems.
  - Explain the kinematically admissible displacement field term as being used in FEM.
  - Explain plain stress condition with one example.
  - Draw eight noded quadrilateral element in Cartesian and natural coordinate systems.
  - Explain Lagrange's shape functions/elements.
  - What is carried out in the processing module of FEM software?
  - What is super-parametric type of Finite Element formulation?
  - Write the equation of 2-D heat conduction problem in terms of temperature ( $T$ ), thermal conductivity ( $k$ ) and heat source/sink ( $Q$ ).
  - Differentiate between complex and multiplex elements with examples.

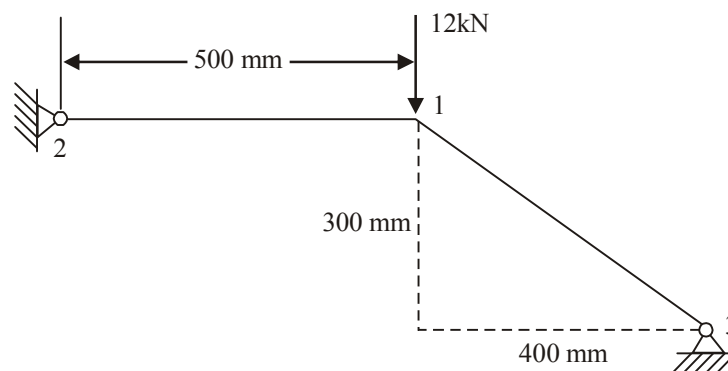
### SECTION-B

- Explain the principle of minimum potential energy and virtual work with their relevant equations.
- Derive the transformation matrix for a 2-D truss element which transforms elemental local displacement to elemental global displacement.
- Explain convergence in reference to Finite Element Method. Also explain the different conditions to achieve the same.
- Derive the stress-strain matrix for plane stress conditions from the basic concept.
- For 1-D bar element, transformation is given as  $\xi = \left( \frac{2}{x_2 - x_1} (x - x_1) - 1 \right)$  which is used to relate  $x$  and  $\xi$ . Let the displacement field is interpolated as  $u(\xi) = N_1 q_1 + N_2 q_2$  where  $N_1 = \cos\left(\frac{\pi(1+\xi)}{4}\right)$  and  $N_2 = \cos\left(\frac{\pi(1-\xi)}{4}\right)$ . Plot the shape functions and develop the strain-displacement matrix. Also develop elemental matrix (you need not to evaluate the integrals).

### SECTION-C

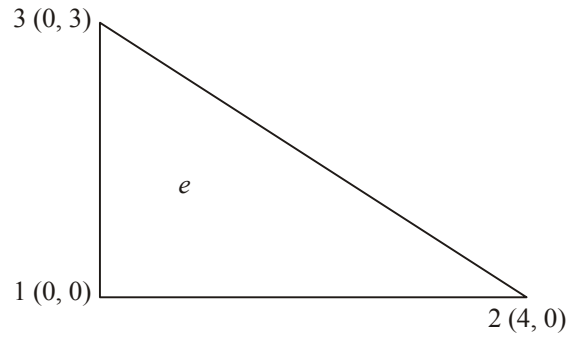
- Explain the Finite Element modeling and shape function for linear interpolation of temperature field (1-D heat transfer element).
- For the two-bar truss as shown in Fig. 1, determine the displacement of node 1. Assume for both members  $E = 70$  GPa and Area = 200 mm<sup>2</sup>. Assume

$$k = \frac{A_e E_e}{L_e} \begin{bmatrix} l^2 & lm & -l^2 & -lm \\ lm & m^2 & -lm & -m^2 \\ -l^2 & -lm & l^2 & lm \\ -lm & -m^2 & lm & m^2 \end{bmatrix} \quad \text{where } l \text{ and } m \text{ have usual meanings.}$$



**Fig. 1**

9. A Constant Strain Triangle (CST) element is shown in **Fig. 2**. The element is subjected to a body force  $f_x = x^2$  N/cm<sup>3</sup>. Determine the nodal force vector due to the body force. Assume element thickness as 1 cm. The coordinates in the **Fig. 2** are given in cm.



**Fig. 2**

**NOTE : Disclosure of Identity by writing Mobile No. or Making of passing request on any page of Answer Sheet will lead to UMC against the Student.**