

**Q. 1 – Q. 5 carry one mark each.**

- Q.1 “The dress \_\_\_\_\_ her so well that they all immediately \_\_\_\_\_ her on her appearance.”

The words that best fill the blanks in the above sentence are

- (A) complemented, complemented      (B) complimented, complemented  
(C) complimented, complimented      (D) complemented, complimented

- Q.2 “The judge’s standing in the legal community, though shaken by false allegations of wrongdoing, remained \_\_\_\_\_.”

The word that best fills the blank in the above sentence is

- (A) undiminished      (B) damaged      (C) illegal      (D) uncertain

- Q.3 Find the missing group of letters in the following series:  
BC, FGH, LMNO, \_\_\_\_\_

- (A) UVWXY      (B) TUVWX      (C) STUVW      (D) RSTUV

- Q.4 The perimeters of a circle, a square and an equilateral triangle are equal. Which one of the following statements is true?

- (A) The circle has the largest area.  
(B) The square has the largest area.  
(C) The equilateral triangle has the largest area.  
(D) All the three shapes have the same area.

- Q.5 The value of the expression  $\frac{1}{1+\log_u vw} + \frac{1}{1+\log_v wu} + \frac{1}{1+\log_w uv}$  is \_\_\_\_\_.

- (A) -1      (B) 0      (C) 1      (D) 3

**Q. 6 – Q. 10 carry two marks each.**

- Q.6 Forty students watched films A, B and C over a week. Each student watched either only one film or all three. Thirteen students watched film A, sixteen students watched film B and nineteen students watched film C. How many students watched all three films?

- (A) 0      (B) 2      (C) 4      (D) 8

Q.7 A wire would enclose an area of  $1936 \text{ m}^2$ , if it is bent into a square. The wire is cut into two pieces. The longer piece is thrice as long as the shorter piece. The long and the short pieces are bent into a square and a circle, respectively. Which of the following choices is closest to the sum of the areas enclosed by the two pieces in square meters?

- (A) 1096                      (B) 1111                      (C) 1243                      (D) 2486

Q.8 A contract is to be completed in 52 days and 125 identical robots were employed, each operational for 7 hours a day. After 39 days, five-seventh of the work was completed. How many additional robots would be required to complete the work on time, if each robot is now operational for 8 hours a day?

- (A) 50                      (B) 89                      (C) 146                      (D) 175

Q.9 A house has a number which needs to be identified. The following three statements are given that can help in identifying the house number.

- i. If the house number is a multiple of 3, then it is a number from 50 to 59.
- ii. If the house number is NOT a multiple of 4, then it is a number from 60 to 69.
- iii. If the house number is NOT a multiple of 6, then it is a number from 70 to 79.

What is the house number?

- (A) 54                      (B) 65                      (C) 66                      (D) 76

- Q.10 An unbiased coin is tossed six times in a row and four different such trials are conducted. One trial implies six tosses of the coin. If H stands for head and T stands for tail, the following are the observations from the four trials:  
(1) HTH THT (2) TTHHHT (3) HTTHHT (4) HHHT\_\_ \_\_.

Which statement describing the last two coin tosses of the fourth trial has the highest probability of being correct?

- (A) Two T will occur.
- (B) One H and one T will occur.
- (C) Two H will occur.
- (D) One H will be followed by one T.

**END OF THE QUESTION PAPER**

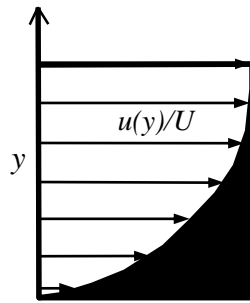
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**Q. 1 – Q. 25 carry one mark each.**

- Q.1 Let  $\vec{a}, \vec{b}$  be two distinct vectors that are not parallel. The vector  $\vec{c} = \vec{a} \times \vec{b}$  is
- (A) zero. (B) orthogonal to  $\vec{a}$  alone.  
(C) orthogonal to  $\vec{a} + \vec{b}$ . (D) orthogonal to  $\vec{b}$  alone.
- Q.2 Consider the function  $f(x, y) = \frac{x^2}{2} + \frac{y^2}{3} - 5$ . All the roots of this function
- (A) form a finite set of points.  
(B) lie on an elliptical curve.  
(C) lie on the surface of a sphere.  
(D) lie on a hyperbolic curve.
- Q.3 Consider a vector field given by  $x\hat{i} + y\hat{j} + z\hat{k}$ . This vector field is
- (A) divergence-free and curl-free.  
(B) curl-free but not divergence-free.  
(C) divergence-free but not curl-free.  
(D) neither divergence-free nor curl-free.
- Q.4 A jet aircraft is initially flying steady and level at its maximum endurance condition. For the aircraft to fly steady and level, but faster at the same altitude, the pilot should
- (A) increase thrust alone.  
(B) increase thrust and increase angle of attack.  
(C) increase thrust and reduce angle of attack.  
(D) reduce angle of attack alone.
- Q.5 The pilot of a conventional airplane that is flying steady and level at some altitude, deflects the port side aileron up and the starboard aileron down. The aircraft will then
- (A) pitch, nose up.  
(B) roll with the starboard wing up.  
(C) pitch, nose down.  
(D) roll with the port wing up.
- Q.6 A NACA 0012 airfoil has a trailing edge flap. The airfoil is operating at an angle of attack of 5 degrees with un-deflected flap. If the flap is now deflected by 5 degrees downwards, the  $C_L$  versus  $\alpha$  curve
- (A) shifts right and slope increases.  
(B) shifts left and slope increases.  
(C) shifts left and slope stays the same.  
(D) shifts right and slope stays the same.

- Q.7 An airplane requires a longer ground roll to lift-off on hot summer days because
- (A) the thrust is directly proportional to free-stream density.
  - (B) the thrust is directly proportional to weight of the aircraft.
  - (C) the lift-off distance is directly proportional to free-stream density.
  - (D) the runway friction is high on hot summer days.

- Q.8 The velocity profile in an incompressible, laminar boundary layer is shown in the figure below.  $U$  is the free-stream velocity,  $u(y)$  is the stream-wise velocity component. The area of the black shaded region in the figure below represents the



- (A) boundary layer thickness.
  - (B) momentum thickness.
  - (C) displacement thickness.
  - (D) shape factor.
- Q.9 The tangential velocity component ' $V$ ' of a spacecraft, which is in a circular orbit of radius ' $R$ ' around a spherical Earth ( $\mu = GM \rightarrow$  gravitational parameter of Earth) is given by the following expression.

(A)  $V = \sqrt{\frac{\mu}{2R}}$       (B)  $V = \sqrt{\frac{\mu}{R}}$       (C)  $V = \frac{2\pi}{\sqrt{\mu}} R^{\frac{3}{2}}$       (D)  $V = \frac{2\pi}{\sqrt{\mu}} R^{\frac{2}{3}}$

- Q.10 Equation of the trajectory of a typical space object around any planet, in polar coordinates ( $r, \theta$ ) (i.e. a general conic section geometry), is given as follows. ( $h$  is angular momentum,  $\mu$  is gravitational parameter,  $e$  is eccentricity,  $r$  is radial distance from the planet center,  $\theta$  is angle between vectors  $\vec{e}$  and  $\vec{r}$ ).

(A)  $r = \frac{(h^2/\mu)}{1-e \cos \theta}$       (B)  $r = \frac{(h^2/\mu)}{e-\cos \theta}$

(C)  $r = \frac{(h^2/\mu)}{1+e \cos \theta}$       (D)  $r = \frac{(h^2/\mu)}{e+\cos \theta}$

- Q.11 In an elliptic orbit around any planet, the location at which a spacecraft has the maximum angular velocity is
- (A) apoapsis.
  - (B) periapsis.
  - (C) a point at  $+45^\circ$  from periapsis.
  - (D) a point at  $-90^\circ$  from apoapsis.

- Q.12 The pitching moment of a positively cambered NACA airfoil about its leading edge at zero-lift angle of attack is
- (A) negative.
  - (B) positive.
  - (C) indeterminate.
  - (D) zero.
- Q.13 In a low-speed wind tunnel, the angular location(s) from the front stagnation point on a circular cylinder where the static pressure equals the free-stream static pressure, is
- (A)  $\pm 38^\circ$
  - (B)  $\pm 30^\circ$
  - (C)  $\pm 60^\circ$
  - (D)  $0^\circ$
- Q.14 A thermocouple, mounted flush in an insulated flat surface in a supersonic laminar flow of air measures the
- (A) static temperature.
  - (B) temperature greater than static but less than total temperature.
  - (C) total temperature.
  - (D) temperature greater than total temperature.
- Q.15 A shock wave is moving into still air in a shock tube. Which one of the following happens to the air?
- (A) static temperature increases, total temperature remains constant.
  - (B) static temperature increases, total temperature increases.
  - (C) static temperature increases, total temperature decreases.
  - (D) static pressure increases, total temperature remains constant.
- Q.16 The highest limit load factor experienced by a civil transport aircraft is in the range
- (A) 0.0 – 2.0
  - (B) 2.0 – 5.0
  - (C) 5.0 – 8.0
  - (D) 8.0 – 10.0
- Q.17 Determine the correctness or otherwise of the following statements, [a] and [r]:
- [a] A closed-section box beam configuration is used in aircraft wings.
- [r] Closed-section box beam configuration is capable of resisting torsional loads.
- (A) Both [a] and [r] are true and [r] is the correct reason for [a].
  - (B) Both [a] and [r] are true but [r] is not the correct reason for [a].
  - (C) Both [a] and [r] are false.
  - (D) [a] is true but [r] is false.

Q.18 The first law of thermodynamics is also known as conservation of

- (A) mass.
- (B) momentum.
- (C) energy.
- (D) species.

Q.19 In an ideal gas turbine cycle, the expansion in a turbine is represented by

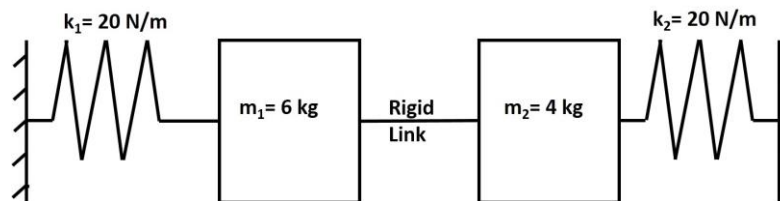
- (A) an isenthalpic process.
- (B) an isentropic process.
- (C) an isobaric process.
- (D) an isochoric process.

Q.20 The determinant of the matrix  $\begin{bmatrix} 1 & 1 & -1 \\ 2 & 1 & 0 \\ 3 & 1 & 1 \end{bmatrix}$  is \_\_\_\_\_ (accurate to one decimal place).

Q.21 The theoretical maximum velocity (in m/s) of air expanding from a reservoir at 700 K is \_\_\_\_\_ (accurate to two decimal places). Specific heat of air at constant pressure is 1005 J/(kg-K).

Q.22 For a damped single degree of freedom system with damping ratio of 0.1, ratio of two successive peak amplitudes of free vibration is \_\_\_\_\_ (accurate to two decimal places).

Q.23 The natural frequency (in rad/s) of the spring-mass system shown in the figure below is \_\_\_\_\_ (accurate to one decimal place).

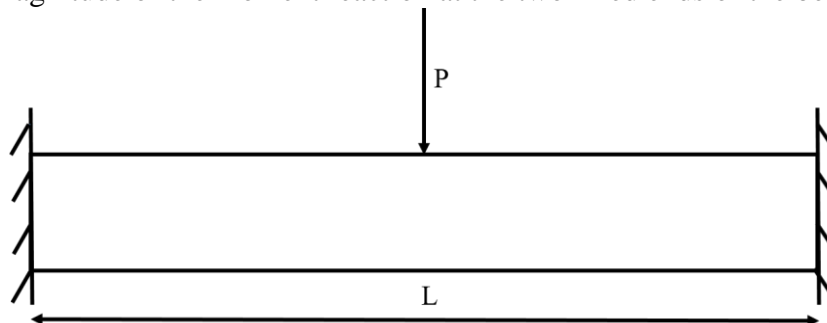


Q.24 The stagnation pressures at the inlet and exit of a subsonic intake are 100 kPa and 98 kPa, respectively. The pressure recovery of this intake will be \_\_\_\_\_ (accurate to two decimal places).

Q.25 A combustor is operating with a fuel-air ratio of 0.03. If the stoichiometric fuel-air ratio of the fuel used is 0.06, the equivalence ratio of the combustor will be \_\_\_\_\_ (accurate to two decimal places).

**Q. 26 – Q. 55 carry two marks each.**

- Q.26 The solution of the differential equation  $\frac{d^2y}{dx^2} + 3\frac{dy}{dx} = 0$ , given that  $y = 0$  and  $\frac{dy}{dx} = 1$  at  $x = 0$  is
- (A)  $x(1 - e^{-3x})$  (B)  $\frac{1}{3}(1 - e^{-3x})$  (C)  $\frac{1}{3}(1 + e^{-3x})$  (D)  $\frac{1}{3}xe^{\frac{-3x}{2}}$
- Q.27 The relation between pressure ( $p$ ) and velocity ( $V$ ) for a steady, isentropic flow at two points along a streamline is, ( $c$  is a constant)
- (A)  $c(p_2^\gamma - p_1^\gamma) = \frac{V_1^2}{2} - \frac{V_2^2}{2}$   
 (B)  $c(p_2^{\frac{\gamma}{\gamma-1}} - p_1^{\frac{\gamma}{\gamma-1}}) = \frac{V_1^2}{2} - \frac{V_2^2}{2}$   
 (C)  $c(p_2^{\frac{\gamma-1}{\gamma}} - p_1^{\frac{\gamma-1}{\gamma}}) = \frac{V_1^2}{2} - \frac{V_2^2}{2}$   
 (D)  $c(p_2^{\gamma-1} - p_1^{\gamma-1}) = \frac{V_1^2}{2} - \frac{V_2^2}{2}$
- Q.28 A thin airfoil is mounted in a low-speed, subsonic wind tunnel, in which the Mach number is 0.1. At a point on the airfoil, the pressure coefficient is measured to be  $-1.2$ . If the flow velocity is increased such that the free-stream Mach number is 0.6, the pressure coefficient at the same point on the airfoil will approximately be:
- (A)  $-3.5$  (B)  $-2.9$  (C)  $-1.5$  (D)  $-0.75$
- Q.29 A solid circular shaft of diameter  $d$  is under pure torsion of magnitude  $T$ . The maximum tensile stress experienced at any point on the shaft is
- (A)  $\frac{32T}{\pi d^3}$  (B)  $\frac{16T}{\pi d^4}$  (C)  $\frac{32T}{\pi d^4}$  (D)  $\frac{16T}{\pi d^3}$
- Q.30 A clamped-clamped beam, subjected to a point load  $P$  at the midspan, is shown in the figure below. The magnitude of the moment reaction at the two fixed ends of the beam is



- (A)  $PL/2$  (B)  $PL/4$  (C)  $PL/8$  (D)  $PL/16$



- Q.31 Which of the following statement(s) is/are true about the state of a body in plane strain condition?  
 P: All the points in the body undergo displacements in one plane only, for example the x-y plane, leading to  $\varepsilon_{zz} = \gamma_{xz} = \gamma_{yz} = 0$ .  
 Q: All the components of stress perpendicular to the plane of deformation, for example the x-y plane, of the body are equal to zero, i.e.  $\sigma_{zz} = \tau_{xz} = \tau_{yz} = 0$ .  
 R: Except the normal component, all the other components of stress perpendicular to the plane of deformation of the body, for example the x-y plane, are equal to zero, i.e.  $\sigma_{zz} \neq 0$ ,  $\tau_{xz} = \tau_{yz} = 0$ .
- (A) P only                      (B) Q only                      (C) P and Q                      (D) P and R
- Q.32 An aircraft with a turbojet engine flies at a velocity of 100 m/s. If the jet exhaust velocity is 300 m/s, the propulsive efficiency of the engine, assuming a negligible fuel-air ratio, is
- (A) 0.33                      (B) 0.50                      (C) 0.67                      (D) 0.80
- Q.33 An aircraft with a turboprop engine produces a thrust of 500 N and flies at 100 m/s. If the propeller efficiency is 0.5, the shaft power produced by the engine is
- (A) 50 kW                      (B) 100 kW  
 (C) 125 kW                      (D) 500 kW
- Q.34 An axial compressor that generates a stagnation pressure ratio of 4.0, operates with inlet and exit stagnation temperatures of 300 K and 480 K, respectively. If the ratio of specific heats ( $\gamma$ ) is 1.4, the isentropic efficiency of the compressor is
- (A) 0.94                      (B) 0.81  
 (C) 0.72                      (D) 0.63
- Q.35 A rocket has an initial mass of 150 kg. After operating for a duration of 10 s, its final mass is 50 kg. If the acceleration due to gravity is  $9.81 \text{ m/s}^2$  and the thrust produced by the rocket is 19.62 kN, the specific impulse of the rocket is
- (A) 400 s                      (B) 300 s  
 (C) 200 s                      (D) 100 s
- Q.36 Consider the vector field  $\vec{v} = -\frac{y}{r^2} \hat{i} + \frac{x}{r^2} \hat{j}$ ; where  $r = \sqrt{x^2 + y^2}$ . The contour integral  $\oint \vec{v} \cdot \vec{ds}$ , where  $\vec{ds}$  is tangent to the contour that encloses the origin, is \_\_\_\_\_ (accurate to two decimal places).
- Q.37 The magnitude of the x-component of a unit vector at the point (1, 1) that is normal to equipotential lines of the potential function  $\phi(r) = \frac{1}{r^2+4}$ , where  $r = \sqrt{x^2 + y^2}$ , is \_\_\_\_\_ (accurate to two decimal places).

- Q.38 Assuming ISA standard sea level conditions (288.16 K, density of  $1.225 \text{ kg/m}^3$ ,  $g = 9.81 \text{ m/s}^2$ ,  $R = 287 \text{ J/(kg-K)}$ ), the density (in  $\text{kg/m}^3$ ) of air at Leh, which is at an altitude of 3500 m above mean sea level is \_\_\_\_\_ (accurate to two decimal places).
- Q.39 Consider a cubical tank of side 2 m with its top open. It is filled with water up to a height of 1 m. Assuming the density of water to be  $1000 \text{ kg/m}^3$ ,  $g$  as  $9.81 \text{ m/s}^2$  and the atmospheric pressure to be 100 kPa, the net hydrostatic force (in kN) on the side face of the tank due to the air and water is \_\_\_\_\_ (accurate to two decimal places).
- Q.40 An aircraft with mass of 400,000 kg cruises at 240 m/s at an altitude of 10 km. Its lift to drag ratio at cruise is 15. Assuming  $g$  as  $9.81 \text{ m/s}^2$ , the power (in MW) needed for it to cruise is \_\_\_\_\_ (accurate to two decimal places).
- Q.41 A statically-stable aircraft has a  $C_{L\alpha} = 5$  (where the angle of attack,  $\alpha$ , is measured in radians). The coefficient of moment of the aircraft about the center of gravity is given as  $C_{M,c.g} = 0.05 - 4\alpha$ . The mean aerodynamic chord of the aircraft wing is 1 m. The location (positive towards the nose) of the neutral point of the aircraft from the center of gravity is \_\_\_\_\_ (in m, accurate to two decimal places).
- Q.42 An aircraft with a gross weight of 2000 kg, has a speed of 130 m/s at sea level, where the conditions are: 1 atmosphere (pressure), 288 K (temperature), and  $1.23 \text{ kg/m}^3$  (density). The speed (in m/s) required by the aircraft at an altitude of 9000 m, where the conditions are: 0.31 atmosphere, 230 K, and  $0.47 \text{ kg/m}^3$ , to maintain a steady, level flight is \_\_\_\_\_ (accurate to two decimal places).
- Q.43 A pitot probe on an aircraft in a steady, level flight records a pressure of  $55,000 \text{ N/m}^2$ . The static pressure and density are  $45,280 \text{ N/m}^2$  and  $0.6 \text{ kg/m}^3$ , respectively. The wing area and the lift coefficient are  $16 \text{ m}^2$  and 2, respectively. The wing loading (in  $\text{N/m}^2$ ) on this aircraft is \_\_\_\_\_ (accurate to one decimal place).
- Q.44 A spacecraft forms a circular orbit at an altitude of 150 km above the surface of a spherical Earth. Assuming the gravitational parameter,  $\mu = 3.986 \times 10^{14} \text{ m}^3/\text{s}^2$  and radius of earth,  $R_E = 6,400 \text{ km}$ , the velocity required for the injection of the spacecraft, parallel to the local horizon, is \_\_\_\_\_ (accurate to two decimal places).
- Q.45 Air at 50 kPa pressure and 400 K temperature flows in a duct at Mach 3.0. A part of the flow leaks through an opening on the duct wall into the ambient, where the pressure is 30 kPa. The maximum Mach number achieved in the discharge is \_\_\_\_\_ (accurate to two decimal places). (Ratio of specific heats of air is  $\gamma = 1.4$ ).

- Q.46 Consider a  $20^\circ$  half-angle wedge in a supersonic flow at Mach 3.0 at standard sea-level conditions. If the shock-wave angle on the wedge is  $36^\circ$ , the Mach number of the tangential component of the flow post-shock is \_\_\_\_\_ (accurate to two decimal places).
- Q.47 The boundary layer thickness at the location of a sensor on a flat plate in an incompressible, laminar flow of air is required to be restricted to 1 mm for an effective measurement. If the flow velocity is 20 m/s with 1 bar pressure, 300 K temperature, and  $1.789 \times 10^{-5}$  kg/(m-s) viscosity, the maximum distance (in mm) of the sensor location from the leading edge is \_\_\_\_\_ (accurate to one decimal place).
- Q.48 Gross weight of an airplane is 7000 N, wing area is  $16 \text{ m}^2$ , and the maximum lift coefficient is 2.0. Assuming density at the altitude as  $1.23 \text{ kg/m}^3$ , the stall speed (in m/s) of the aircraft is \_\_\_\_\_ (accurate to two decimal places).
- Q.49 A thin-walled tube with external radius of 100 mm and wall thickness of 2 mm, is fixed at one end. It is subjected to a compressive force of 1 N acting at a point on the circumference parallel to its length. The maximum normal stress (in kPa) experienced by the structure is \_\_\_\_\_ (accurate to two decimal places).
- Q.50 A 1 m long massless cantilever beam oscillates at 2Hz, while a 60 kg mass is attached at the tip of it. The flexural rigidity of the beam (in  $\text{kN-m}^2$ ) is \_\_\_\_\_ (accurate to two decimal places).
- Q.51 A cantilever beam having a rectangular cross-section of width 60 mm and depth 100 mm, is made of aluminum alloy. The material mechanical properties are: Young's modulus,  $E = 73 \text{ GPa}$  and ultimate stress,  $\sigma_u = 480 \text{ MPa}$ . Assuming a factor of safety of 4, the maximum bending moment (in  $\text{kN-m}$ ) that can be applied on the beam is \_\_\_\_\_ (accurate to one decimal place).
- Q.52 The components of stress in a body under plane stress condition, in the absence of body forces, is given by:  
 $\sigma_{xx} = Ax^2$ ;  $\sigma_{yy} = 12x^2 - 6y^2$  and  $\sigma_{xy} = 12xy$ .  
The coefficient, A, such that the body is under equilibrium is \_\_\_\_\_ (accurate to one decimal place).
- Q.53 An axial compressor rotor with 50 % degree of reaction, operates with an axial velocity of 200 m/s. The absolute flow angle at the inlet of the rotor is  $22^\circ$  with reference to the axial direction. If the axial velocity is assumed to remain constant through the rotor, the magnitude of the relative velocity (in m/s) at the rotor exit is \_\_\_\_\_ (accurate to one decimal place).

- Q.54 The relative velocity of air leaving a straight radial impeller of a centrifugal compressor is 100 m/s. If the impeller tip speed is 200 m/s, for a slip free operation, the absolute velocity (in m/s) at the impeller exit is \_\_\_\_\_ (accurate to one decimal place).
- Q.55 An aircraft wind tunnel model, having a pitch axis mass moment of inertia ( $I_{yy}$ ) of 0.014 kg-m<sup>2</sup>, is mounted in such a manner that it has pure pitching motion about its centre of gravity, where it is supported through a frictionless hinge. If the pitching moment (M) derivative with respect to angle of attack ( $\alpha$ ), denoted by ' $M_\alpha$ ', is -0.504 N-m/rad and the pitching moment (M) derivative with respect to pitch rate ( $q$ ), denoted by ' $M_q$ ', is -0.0336 N-m/(rad/s), the damping ratio of the resulting motion due to an initial disturbance in pitch angle is approximately \_\_\_\_\_ (accurate to three decimal places).

**END OF THE QUESTION PAPER**

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