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## **GUJARAT TECHNOLOGICAL UNIVERSITY**

**BE - SEMESTER-VII(NEW) EXAMINATION - SUMMER 2019** 

Subject Code:2170107 Date:10		6/05/2019 Iarks: 70	
Subject Name: Mechanics of Composite Materials Time: 02:30 PM TO 05:00 PM Total N Instructions:			
	<ol> <li>Att</li> <li>Ma</li> <li>Fig</li> </ol>	empt all questions. ke suitable assumptions wherever necessary. ures to the right indicate full marks.	
			MARKS
Q.1	( <b>a</b> )	What is the limitation of a unidirectional lamina? How it can be overcome by laminate? Explain with neat sketch.	03
	<b>(b</b> )	Define composite material. Explain basic constituent materials of composites	04
	( <b>c</b> )	Carry out the comparison of composite material with metals in detail.	07
Q.2	(a)	Draw a neat sketch of variation of stress and strain through the thickness of the laminate.	03
	<b>(b)</b>	The E-glass fibres in a polyester resin is 35% by weight. Given $\rho_f = 2.50$ g/ml and $\rho_m = 1$ g/ml. Calculate $V_f$ and $\rho_c$ for the lamina.	04
	(c)	Identify the types of laminate given below. 1. $[0/30/60]_s$ 2. $[0/90_2/0/90]$ 3. $[-40/40/-40/40]$ 4. $[45/60/-60/-45]$ 5. $[30/40/-30/30/-30/-40]$ 6. $[0/\pm 60]$ 7. $[0/90]_s$ OR	07
	(c)	Evaluate the transverse modulus $E_2$ of a composite lamina by (i) strength of material approach and (ii) Halpin-Tsai relation, with the following properties $E_{2f} = 14.8$ GPa, $E_m = 3.45$ GPa, $\nu_m = 0.36$ , $V_f = 0.65$ . Assume $\xi_1 = 1$	07
Q.3	<b>(a)</b>	Explain about symmetric laminate.	03
	<b>(b</b> )	Derive the equation of Inplane Shear modulus using neat sketchs.	04
	(c)	Derive Strain-Displacement relationship of a laminate using a neat sketch.	07
Q.3	<b>(a)</b>	What do you mean by balanced laminate.	03
	(b)	Derive the equation of Transverse modulus with the use of neat sketches.	04
	( <b>c</b> )	Write down [A], [B] and [D] matrices and explain its significance in detail.	07
Q.4	<b>(a)</b>	Write down the difference between micromechanics and macromechanics.	03



FirstRanker.com Firstrankoj's chojass/epoxy specimeriveinhing er?8 cm was burnt and the second to be 0.49 cm. Densities of glass of the remaining fibres was found to be 0.49 gm. Densities of glass and epoxy are 2.4 gm/ml and 1.20 gm/ml respectively. Determine the density of composites in the absence of voids.

> Discuss the application of composite materials in various 07 (c) engineering fields.

> > OR

- Q.4 Explain about prepegs. **(a)** 
  - Derive the rule of mixtures relation for major Poisson's ratio and **(b)** 04 write expression for minor Poisson's ratio.
  - **(c)** Write down the advantages and disadvantages of composite 07 material.
- Q.5 Reduced stiffness matrix of an orthotropic lamina is given by, **(a)** 03 [150.81 4.027 0]

$$[Q] = \begin{bmatrix} 130.01 & 4.027 & 0\\ 4.027 & 20.11 & 0\\ 0 & 0 & 5 \end{bmatrix}$$
GPa

Determine  $E_1$ ,  $E_2$ ,  $G_{12}$  and  $v_{12}$ .

- Explain the classification of composite materials in detail. **(b)** 04
- How the independent elastic constants have been reduced from 81 to 07 (c) 2 for different materials. Explain in detail with the use of stressstrain relations and neat sketches.

## OR

- 03 Q.5 For an orthotropic lamina, engineering constants along the principal **(a)** material axis are  $E_1 = 150$  GPa,  $E_2 = 20$  GPa,  $G_{12} = 5$  GPa,  $v_{12} = 5$ 0.2. Determine the reduced stiffness matrix [Q].
  - **(b)** Write down the significance of fillers and additives in composite 04 materials in detail.
  - Anisotropic lamina has  $E = 100 \text{ kN/mm}^2$  and  $\nu = 0.25$ . Determine 07 (c) reduced stiffness matrix and reduced compliance matrix.

03