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Total No. of Pages : 02

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# B.Tech.(ANE) (Sem.-5) AIRCRAFT PERFORMANCE Subject Code : ANE-315 M.Code : 60524

Time: 3 Hrs.

Max. Marks: 60

## INSTRUCTIONS TO CANDIDATES :

- 1. SECTION-A is COMPULSORY consisting of TEN questions carrying TWO marks each.
- 2. SECTION-B contains FIVE questions carrying FIVE marks each and students have to attempt any FOUR questions.
- 3. SECTION-C contains THREE questions carrying TEN marks each and students have to attempt any TWO questions.

## **SECTION-A**

#### **1.** Attempt the following :

- a) Define 'Wing loading' and explain its importance.
- b) Distinguish between 'Range' and 'Endurance'.
- c) Define 'Aerodynamic Center' & 'Center of Pressure'
- d) Distinguish between 'Geometric' and 'Geo-potential' altitude.
- e) Distinguish between 'Isothermal' and 'Gradient' regions.
- f) Distinguish between 'Washin' and 'Washout'.
- g) Distinguish between 'Chord line' and 'Camber line'.
- h) Define 'Temperature altitude'.
- i) Define 'Drag divergence Mach number' and explain its importance.
- j) Define 'zero-lift' angle of attack.



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#### **SECTION-B**

- 2. Calculate the standard atmospheric values of T, p and  $\rho$  at a geo-potential altitude of 18 km. Assume Lapse rate of -6.5 K/km in the gradient region. (1,2,2)
- 3. Explain 'Delta Wing Aerodynamics' at Low Speed with the help of labeled illustration/plots. (5)
- 4. Derive the expression for lift coefficient and drag coefficient for minimum power condition. (5)
- 5. Define and explain various high lift devices. Show their effect on lift curve slopes. (5)
- 6. Derive the expression for take-off distance. (5)

#### **SECTION-C**

- 7. Gulfstream-IV twin turbofan executive transport with weight of 200000N, planform area of 80 m<sup>2</sup> and drag polar as  $C_D = 0.015 + 0.08C_L^2$  is flying at an altitude of 10 km ( $\rho = 0.413 \text{ kg/m}^3$ ). Calculate :
  - a)  $(C_L^{3/2}/C_D)_{max}$ ,  $(C_L/C_D)_{max}$  and  $(C_L^{1/2}/C_D)_{max}$  values. (5)
  - b) Velocities at which  $(C_L^{3/2}/C_D)_{max}$ ,  $(CL/C_D)_{max}$  and  $(C_L^{1/2}/C_D)_{max}$  occur. (5)
- 8. Write notes on the following :
  - a) Drag and its categorization (5)
  - b) V/STOL vehicles (5)
- 9. Define 4-digit, 5-digit and 6-digit NACA series airfoils. What are laminar and supercritical airfoils? Where are these airfoils used? (6, 4)

# NOTE : Disclosure of Identity by writing Mobile No. or Making of passing request on any page of Answer Sheet will lead to UMC against the Student.