

**Total No. of Pages : 02**

**Total No. of Questions : 09**

**B.Tech.(ANE) (Sem.-6)**

# ELEMENTS OF SPACECRAFT ENGINEERING

**Subject Code : ANE-323**

**M.Code : 60532**

**Time : 3 Hrs.**

**Max. Marks : 60**

**INSTRUCTIONS TO CANDIDATES :**

1. **SECTION-A is COMPULSORY consisting of TEN questions carrying TWO marks each.**
2. **SECTION-B contains FIVE questions carrying FIVE marks each and students have to attempt any FOUR questions.**
3. **SECTION-C contains THREE questions carrying TEN marks each and students have to attempt any TWO questions.**

## SECTION-A

1.
  - a) What is a geostationary satellite?
  - b) Name various types of fuels and oxidizers used in solid propellants.
  - c) What are functions of missiles?
  - d) What is the role of strap-on boosters?
  - e) Why multi-stage rockets are preferred over single stage rocket?
  - f) Define 3rd Law of Kepler's planetary motion.
  - g) Name six elements which completely define an orbit.
  - h) Define orbital velocity.
  - i) What is the effect of earth oblateness on satellite orbit?
  - j) What is prolate satellite?

**SECTION-B**

2. With the help of a neat diagram, explain construction, working and salient features of a liquid rocket engine.
3. Derive Tsiolkovsky's equation for the rocket in terms of mass ratio and specific impulse of the rocket.
4. With the help of a diagram, show all the orbital elements and state and prove Kepler's 2<sup>nd</sup> law of planetary motion.
5. Explain the boost phase and ballistic phase of the trajectory of a ballistic missile.
6. A satellite in low earth orbit starts falling due to atmospheric drag. Derive an expression to estimate the time when satellite hits the ground.

**SECTION-C**

7. With the help of neat diagrams explain the methods for attitude control of spinning and non-spinning spacecraft.
8. The data for a two stage rocket is given as below :  
  
First stage : propellant mass = 3450 kg, structure mass = 400 kg.  
  
Second stage : propellant mass = 1000 kg, structure mass = 100 kg.  
  
The payload mass = 50 kg and specific impulse is 350 seconds. Calculate the burnout velocity of the rocket.
9. Write notes on :
  - a) Parallel staging of rockets 3
  - b) Inclination change maneuver 3
  - c) Earth oblateness effect 4

**NOTE : Disclosure of identity by writing mobile number or making passing request on any page of Answer sheet will lead to UMC against the Student.**