Code No: RT31035 (R13) (SET - 1)

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**R13** 

SET - 1

Discuss the relative advantages and disadvantages of gas turbines and steam turbines. 6 a)

[6M] [10M]

[9M]

A simple turbine jet unit was tested when stationary and the ambient conditions were 1bar and 15<sup>o</sup>C. The pressure ratio for the compressor was 4:1. A fuel consumption of 0.37kg/s was obtained for an air flow of 23kg/s. Calculate the thrust produced if the exhaust gases from the turbine were expanded to atmospheric pressure in a convergent nozzle. Assume the following data:

Isentropic efficiency of compressor-80%

Isentropic efficiency of turbine-85%

Efficiency of nozzle-93%

Transmission efficiency-98%

Calorific value of fuel-42000kJ/kg

Assuming working fluid to be air throughout.

Describe with a suitable sketch the constant pressure open cycle gas turbine. 7 a) [7M]

Derive expressions for the thrust and propulsion efficiency of rockets and compare with b) those of turbojet.

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