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Total No. of Pages : 03

Total No. of Questions : 09

B.Tech. (ANE) (Sem.-6)
AUTOMATIC FLIGHT CONTROL
Subject Code : ANE-321
Paper ID : [A1227]

Time : 3 Hrs.

Max. Marks : 60

INSTRUCTIONS TO CANDIDATES :

1. SECTION-A is COMPULSORY consisting of TEN questions carrying TWO marks each.
2. SECTION-B contains FIVE questions carrying FIVE marks each and students have to attempt any FOUR questions.
3. SECTION-C contains THREE questions carrying TEN marks each and students have to attempt any TWO questions.

SECTION-A**1. Answer briefly :**

- a) Distinguish between negative feedback vs. positive feedback closed loop system.
- b) What is open loop control system? Explain with example.
- c) How does the location of poles of a transfer function effects the stability?
- d) Draw the block diagram of closed loop unity feedback system, clearly labeling transfer function, feedback control function, actuator dynamics and reference input.
- e) Give example of a system, *i.e.* give transfer function which can be stabilized using integral control.
- f) Explain the condition under which a polynomial is called 'Hurwitz'.
- g) What is peak time and rise time of a system?
- h) Draw block diagram of displacement autopilot.
- i) Explain the relation between automatic flight controls and instrument landing system.
- j) How coordination can be obtained in eliminating sideslip turn?

SECTION-B

2. Distinguish between following types of systems, along with a typical governing equation of each type :

- a) SISO and MIMO.
- b) Linear and non-linear system.

3. Consider a first order system with governing equation as :

$$A \frac{dy(t)}{dt} + y(t) = u(t)$$

Design a system with settling time of 2 seconds.

4. Transfer function of liner time invariant system is $\frac{s+2}{s+3}$. Calculate magnitude and phase of sinusoidal transfer function at frequency 2 rad/s.
5. The open loop transfer function of negative feedback closed loop system is :

$$\frac{k(s+4)}{(s^2+4s-6)}$$

Where k is non-negative real valued parameter.

- a) How many asymptotes will be there?
 - b) The point of intersection between asymptotes, (calculate all such points) Comment on break-in and break-out points if any.
6. In context of yaw orientation control system answer the following :
- a) Draw block diagram of the yaw orientation control system for root locus analysis.
 - b) Derive expression of transfer function of the same.

SECTION-C

7. A closed loop control with unity negative feedback system having governing equation as given below is designed using proportional controller :

$$\frac{d^2y(t)}{dt^2} + 4 \frac{dy(t)}{dt} - 6y(t) = u(t)$$

- a) For what value of K_p above system will be stable.
- b) Calculate the steady state tracking error for unity step reference input.

8. For the given open loop transfer function :

$$G(s)H(s) = \frac{k}{(s+3)(s-1)}$$

Calculate :

- a) Locate the parts of the real axis that lies on the real axis.
 - b) Determine the asymptotes of the root loci.
 - c) Determine the break-away and break-in points.
 - d) Sketch the root locus.
9. In context of pitch orientation control system answer the following :
- a) Draw root locus for complete pitch orientation control system for low angles of attack.
 - b) Draw root locus for complete pitch orientation control system for high angles of attack.
 - c) Explain major differences between the same.